

# An Omnidirectional Dual-Reflector Antenna with a Shaped Main Reflector Described by Local Conic Sections

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**Abstract**— This work investigates the synthesis of dual-reflector antennas for omnidirectional coverage. Both reflectors are bodies-of-revolution: the subreflector is generated by a single conic section while the shaped main reflector is generated by a series of local conic sections which are consecutively concatenated. The main-reflector is shaped to radiate, under GO principles, a prescribed power density in the elevation plane. To illustrate the method, two representative omnidirectional axis-displaced Cassegrain configurations are synthesized. The GO shaping results are validated by method-of-moments simulations.

## I. INTRODUCTION

The growing demand for broadband Internet access as well as the increasing quality and speed of communication links have set an important niche market known as “last mile” access. WiMax systems, which are based on the cellular concept, may be employed in these services. Furthermore, by operating in the microwave and millimetre-wave ranges, one may employ reflector antennas while planning cells with omnidirectional coverage. The main advantages of this kind of antennas are their potentially high gain and relatively broad operation band.

Omnidirectional dual-reflector antennas with a shaped main reflector may be synthesized to provide a prescribed radiation pattern in the elevation plane [1]. Some reflector shaping techniques are based on the numerical integration of an ordinary differential equation derived from Geometrical Optics (GO) principles [1]. Recently, a technique based on the combination of local conic sections has been presented for the shaping of Cassegrain and Gregorian antennas [2]. The formulation was improved [3] and then extended to the shaping of omnidirectional dual-reflector configurations [4]. In [2]-[4], both sub- and main-reflectors are shaped to provide a uniform phase distribution at the main-reflector aperture. In the present work, we extend the concepts of [2]-[4] to the design of omnidirectional dual-reflector antennas where only the main reflector is shaped to provide the desired radiation pattern in the elevation plane. To illustrate the proposed technique, a design similar to that presented in [5] is investigated. It is shown that the present shaping technique

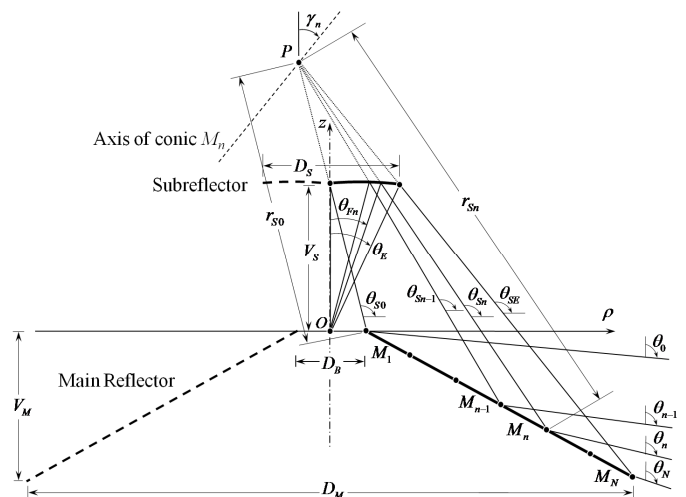


Fig. 1 - Conic sections representing the generatrix shaped main reflector

converges much more rapidly than the technique based on the numerical integration of a differential equation [1].

## II. MAIN-REFLECTOR SHAPING TECHNIQUE

In the present shaping technique, the axis-symmetric subreflector is generated by a conic section, while the main reflector is generated by the consecutive combination of local conic sections. A procedure for the definition of the subreflector conic generatrix is discussed in [1]. This conic has 2 foci: one located at the origin ( $O$ ), where the feed phase center is supposed to be, and another at  $P$ , which defines the subreflector annular caustic (see Fig. 1). As the subreflector generating conic is known, the relation between the feed-ray direction ( $\theta_F$ ) and the direction of the ray reflected by the subreflector ( $\theta_S$ ) is also known [1]. The main reflector is generated by several local conic section  $M_n$  ( $n = 1, \dots, N$ ), all of them with one of their foci at  $P$ . The axis of  $M_n$  passes through  $P$  and makes an angle  $\gamma_n$  with the  $z$ -direction (Fig. 1). In order to uniquely define the main-reflector local conic sections, three parameters must be determined for each  $M_n$ : the inter-focal distance ( $2c_n$ ), the eccentricity ( $e_n$ ), and the tilt

with the help of the following auxiliary parameters [6]:

$$a_n = c_n (e_n - 1/e_n) \quad (1)$$

$$b_n = e_n \sin \gamma_n \quad (2)$$

$$d_n = e_n \cos \gamma_n - 1 \quad (3)$$

Three equations are needed to obtain the parameters  $a_n$ ,  $b_n$ , and  $d_n$  at each step  $n$  of the iterative process. From the polar equation of  $M_n$ , the distance from  $P$  to  $M_n$  is given by:

$$r_s = \frac{a_n}{b_n \sin \theta_s + (1 + d_n) \cos \theta_s - 1}, \quad \text{for } \theta_{s_{n-1}} \leq \theta_s \leq \theta_{s_n} \quad (4)$$

Since  $\theta_{s_{n-1}}$  and  $r_{s_{n-1}}$  are both known from the previous step ( $n-1$ ), from (4) one obtains:

$$r_{s_{n-1}} = \frac{a_n}{b_n \sin \theta_{s_{n-1}} + (1 + d_n) \cos \theta_{s_{n-1}} - 1} \quad (5)$$

Also from the polar equation of  $M_n$  one obtains that

$$b_n \left[ \cot \left( \frac{\theta}{2} \right) + \cot \left( \frac{\theta_s}{2} \right) \right] + d_n \left[ \cot \left( \frac{\theta}{2} \right) \cot \left( \frac{\theta_s}{2} \right) - 1 \right] = 2, \quad (6)$$

for  $\theta_{s_{n-1}} \leq \theta_s \leq \theta_{s_n}$  and  $\theta_{n-1} \leq \theta \leq \theta_n$ , where  $\theta$  is the direction of the ray reflected by the main reflector (Fig. 1). As  $\theta_{s_{n-1}}$  and  $\theta_{n-1}$  are known from the previous step, from (6) one obtains:

$$b_n \left[ \cot \left( \frac{\theta_{n-1}}{2} \right) + \cot \left( \frac{\theta_{s_{n-1}}}{2} \right) \right] + d_n \left[ \cot \left( \frac{\theta_{n-1}}{2} \right) \cot \left( \frac{\theta_{s_{n-1}}}{2} \right) - 1 \right] = 2 \quad (7)$$

Finally, the conservation of energy along the tube of rays departing from  $O$  toward the main-reflector far-field region is imposed:

$$\int_0^{\theta_{fn}} G_F(\theta_F) \sin \theta_F \, d\theta_F = N_F \int_{\theta_0}^{\theta_n} G_A(\theta) \sin \theta \, d\theta \quad (8)$$

where  $G_F(\theta_F)$  and  $G_A(\theta)$  are the circularly-symmetric radiation patterns of the feed and the antenna, respectively,  $\theta_0$  is the prescribed direction of the ray reflected by the main reflector corresponding to the feed principal ray ( $\theta_F = 0$ ), and  $N_F$  is a normalization constant give by

$$N_F = \int_0^{\theta_E} G_F(\theta_F) \sin \theta_F \, d\theta_F / \int_{\theta_0}^{\theta_N} G_A(\theta) \sin \theta \, d\theta \quad (9)$$

where  $\theta_0$  and  $\theta_N$  are the far field limits under GO principles. If  $\theta_0 > \theta_N$  the structure of rays corresponds to a main reflector with real caustic. If  $\theta_N > \theta_0$ , then the caustic is virtual. Once  $\theta_n$  is numerically determined from (8), the third shaping equation is obtained from (6):

$$b_n \left[ \cot \left( \frac{\theta_n}{2} \right) + \cot \left( \frac{\theta_{s_n}}{2} \right) \right] + d_n \left[ \cot \left( \frac{\theta_n}{2} \right) \cot \left( \frac{\theta_{s_n}}{2} \right) - 1 \right] = 2 \quad (10)$$

From (5), (7), and (9) a set of three linear equations are established to determine  $a_n$ ,  $b_n$ ,  $d_n$ , and, with the help of (1)-(3),  $2c_n$ ,  $e_n$ ,  $\gamma_n$ .

The iterative process starts at  $n = 0$ , with  $\theta_{r0} = 0$  ( $\theta_{s0}$  is obtained from  $\theta_{r0}$  with the help of the subreflector conic equation),  $\theta = \theta_0$ , and  $r_{s0}$  given by the location of the subreflector caustic  $P$  and by the desired diameter  $D_B$  of the main-reflector central opening (see Fig. 1).

### III. CASE STUDIES

To illustrate the shaping procedure described in the previous section, two case studies are discussed: the first example is designed to provide a sectoral coverage in the elevation plane over  $\pm 7.5^\circ$  with respect to the horizon, while the second provides a sectoral coverage over  $\pm 15^\circ$ . In both examples the antennas have been scaled to have a conical projected aperture width ( $W_A$ ) equal to  $8.25\lambda$  or  $16.5\lambda$ .

The feed is assumed to be a TEM coaxial horn. The function that describes the radiation pattern of the feeder in the far-field region is given by [1]:

$$G_F(\theta_F) = G_{0F} \left[ \frac{J_0(kr_i \sin \theta_F) - J_0(kr_e \sin \theta_F)}{\sin \theta_F} \right]^2 \quad (11)$$

where  $G_{0F}$  is a normalization factor,  $r_i$  and  $r_e$  are the internal and external radii, respectively. In all case studies to be presented,  $r_i = 0.45\lambda$  and  $r_e = 0.90\lambda$ . The objective function that describes the sectoral radiation pattern of the dual-reflector antenna in the elevation plane is given by

$$G_A(\theta) = \frac{1}{2\pi} \left| \frac{1}{\cos \theta_1 - \cos \theta_2} \right| \quad (12)$$

which has been normalized for unit radiated power.

#### A. OADC Configuration for Sectoral Coverage over $\pm 7.5^\circ$

In the first case study (Case A.I), the main reflector of an omnidirectional axis-displaced Cassegrain (OADC) configuration was synthesized to radiate, under GO principles, a sectoral illumination in the elevation plane over  $\pm 7.5^\circ$  with respect to the horizon (i.e., from  $\theta = \theta_N = 82.5^\circ$  to  $\theta = \theta_0 = 97.5^\circ$ ). As  $\theta_0 > \theta_N$ , the main reflector has a real caustic (see Fig. 2). This is basically the design presented in [5], except that here the feed is simulated as a TEM coaxial horn and the subreflector is generated by an ellipse to avoid rays being reflected back to the feed aperture. That can be noticed from Fig. 2 by observing the direction of the rays reflected by subreflector. By minimizing the scattering back to the feed aperture one may end up improving the return-loss behaviour across the operation bandwidth (such topic is beyond the objectives of the present work).

The subreflector generating ellipse was determined from a classical OADC configuration with the following parameters [1] (some of them are illustrated in Fig. 1):  $V_S = 9.5\lambda$ ,  $D_B = 2.4\lambda$ , main-reflector projected diameter  $D_M = 20\lambda$ , focus  $O$  at the plane of the main-reflector central opening ( $z_B = 0$ ), main-reflector aperture width  $W_A = 8.25\lambda$ , and  $\gamma = 90^\circ$ . These values provide an ellipse with an eccentricity equal to 0.7871, an inter-focal distance equal to  $69.93\lambda$ , and an axis with a tilt

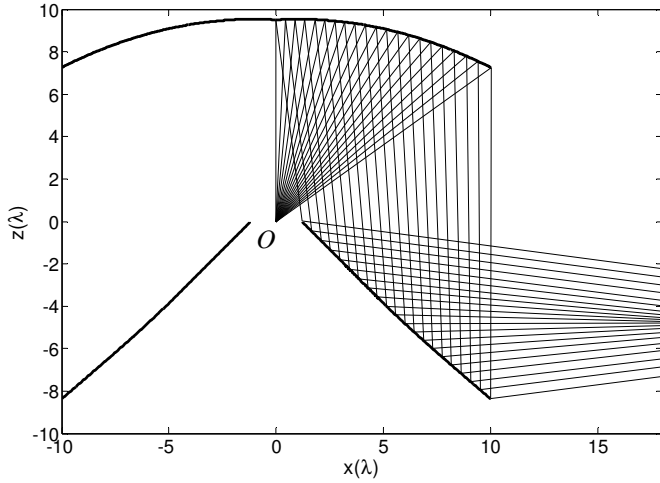


Fig. 2 – Generatrices of the shaped OADC dual-reflector antenna (Case A.I) with sectoral illumination in elevation plane over  $\pm 7.5^\circ$  with respect to horizon.

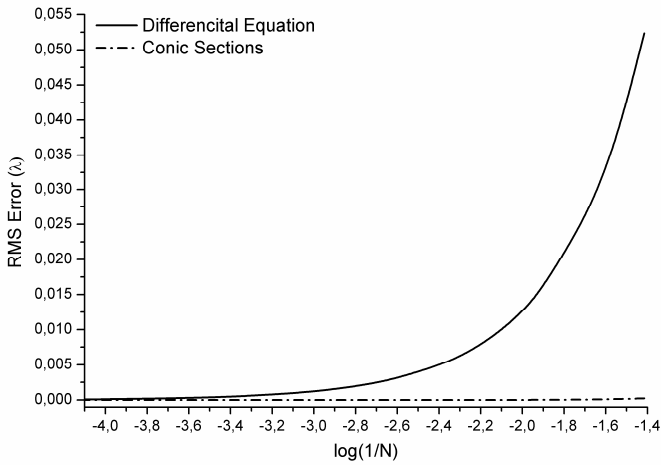


Fig. 3 – RMS error with respect to  $N$  of the main-reflector of Case A.I.

angle of  $171.82^\circ$  with respect to the  $z$ -axis. As a consequence, the projected diameter of the subreflector  $D_S = 20.03\lambda$  and the subreflector edge angle  $\theta_E = 54.07^\circ$ . These parameters also provide  $r_{S0} = -69.77\lambda$  as an initial parameter for the shaping procedure presented in Sect. II. The negative value of  $r_{S0}$  indicates that the ring caustic  $P$  is *below* the main reflector (see Fig. 2). The generatrices of the subreflector and shaped main reflector are illustrated in Fig. 2.

The main-reflector shaping was also performed with the procedure described in [1], which is based on the numerical integration of a differential equation. The main-reflector RMS error (with respect to a reflector shaped with a very large  $N$  value) obtained by the present technique and that of [1] is depicted in Fig. 3 as a function of  $\log(1/N)$ , where  $N+1$  is the total number of points over the main-reflector generatrix, obtained at each step of the shaping procedures (see Fig. 1). Only these  $N+1$  points were considered in the RMS error calculation. The errors refer to  $r_S$ , i.e., the distance from  $P$  to the main reflector. From Fig. 3 one observes that, for this particular design, the present shaping procedure provides the

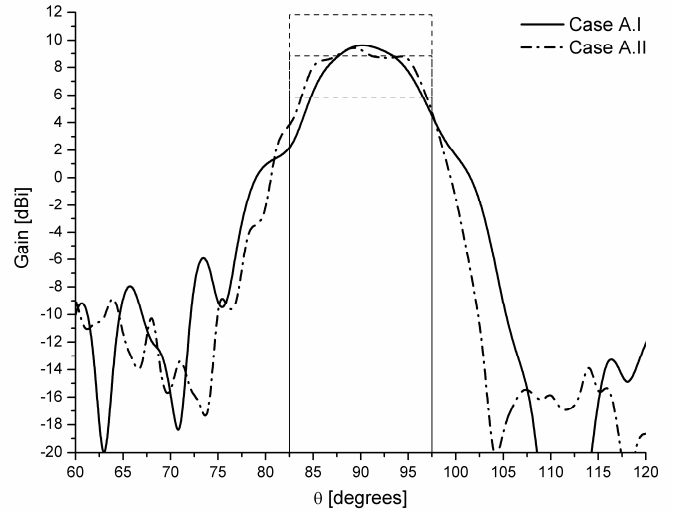


Fig. 4 – MoM radiation patterns (Cases A.I and A.II) in the elevation plane.

TABLE I

| Parameters      | Case A.I      | Case A.II     | Case B.I      | Case B.II     |
|-----------------|---------------|---------------|---------------|---------------|
| $W_A (\lambda)$ | 8.25          | 16.50         | 8.25          | 16.50         |
| $V_S (\lambda)$ | 9.50          | 19.00         | 9.50          | 19.00         |
| $D_M (\lambda)$ | 20.00         | 40.00         | 20.00         | 40.00         |
| $D_B (\lambda)$ | 2.40          | 4.80          | 2.40          | 4.80          |
| $D_S (\lambda)$ | 20.03         | 40.06         | 20.03         | 40.06         |
| $V_M (\lambda)$ | 8.37          | 16.74         | 8.58          | 17.16         |
| $e$             | 0.787098      | 0.787098      | 0.787098      | 0.787098      |
| $\theta_E$      | $54.07^\circ$ | $54.07^\circ$ | $54.07^\circ$ | $54.07^\circ$ |

same accuracy of the procedure presented in [1] using about 100 times less points.

In another situation (Case A.II), the antenna of Case A.I was scaled to have its dimensions doubled. Its dimensions are listed in Table I. It is expected that contributions to the field generated by the diffraction phenomena and electromagnetic coupling are less significant and, consequently, the synthesis based on GO have a better result. The radiation patterns of Cases A.I and A.II were simulated by the method of moments (MoM) and are illustrated in Fig. 4, together with the prescribed GO far-field pattern of (8), used for the main-reflector shaping [7]. As expected, the antenna with a bigger aperture (Case A.II) presents a radiation pattern closer to the objective function than that of Case A.I.

### B. OADC Configuration for Sectoral Coverage over $\pm 15^\circ$

The omnidirectional ADC configuration of Case B.I was designed to radiate, once again, a sectoral illumination in the elevation plane but over a wider range ( $\pm 15^\circ$  with respect to the horizon, i.e., from  $\theta = \theta_N = 75^\circ$  to  $\theta = \theta_0 = 105^\circ$ ). The subreflector is the same of Case A.I and the main parameters are listed in Table I. The generatrices of the subreflector and shaped main reflector are illustrated in Fig. 5. The main-reflector RMS error of Case B.I is illustrated in Fig. 6 and

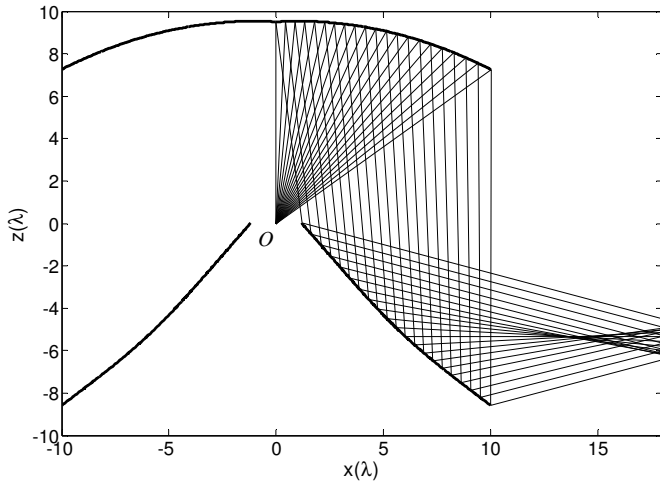


Fig. 5 – Generatrices of the shaped OADC dual-reflector antenna (Case B.I) with sectoral illumination in elevation plane over  $\pm 15^\circ$  with respect to horizon.

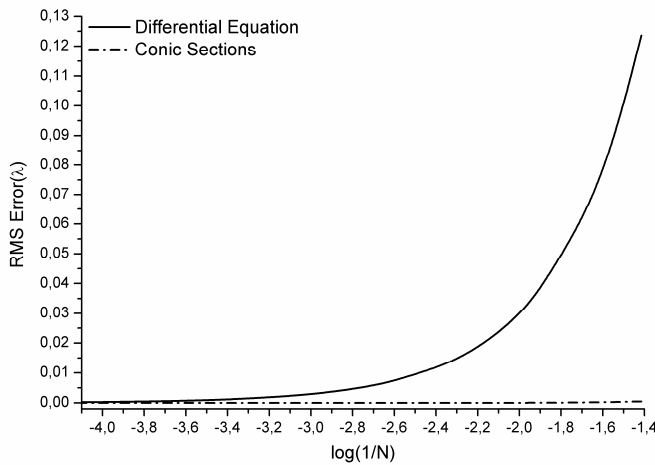


Fig. 6 – RMS error with respect to  $N$  of the main-reflector of Case B.I.

basically has the same behaviour of the RMS error of Case A.I (Fig. 3), differing only in the error scale.

In last situation analysed (Case B.II), the dual-reflector antenna of Case B.I was scaled to have its dimensions doubled (Table I). Figure 7 illustrates the MoM radiation patterns of Cases B.I and B.II, together with the prescribed GO pattern (8). One observes from Fig. 7 that the bigger antenna (Case B.II) provides a radiation pattern closer to the prescribed objective.

#### IV. CONCLUSIONS

This work presented a method for the GO synthesis of omnidirectional dual-reflector antennas. The procedure shapes the main-reflector generatrix, which is represented by local conic sections consecutively concatenated. The subreflector is not shaped and its generatrix is represented by a single conic section. Given the feed illumination, the main-reflector is shaped to radiate a prescribed power density in the elevation plane. The procedure was demonstrated in the synthesis of

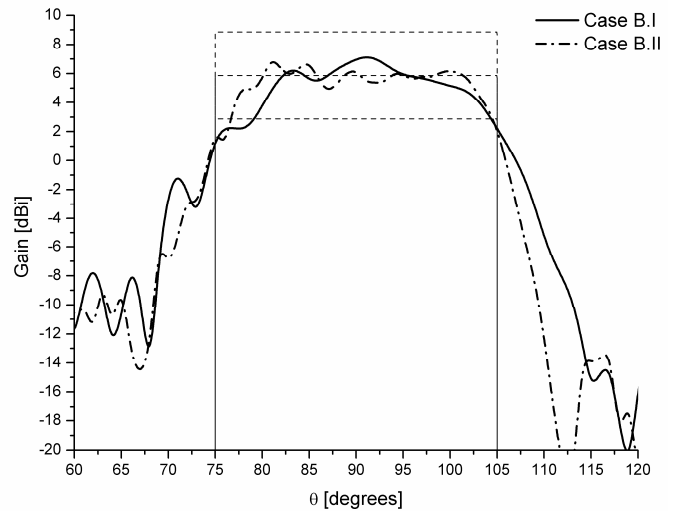


Fig. 7 – MoM radiation patterns (Cases B.I and B.II) in the elevation plane

two representative configurations, shaped to radiate a sectoral power density. For the case studies investigated here, the method has a numerical convergence higher than a traditional method based on the numerical integration of an ordinary differential equation.

Base on GO principles that shape the main reflector of omnidirectional dual-reflector antenna by concatenation of conic sections in other to obtain a uniform radiation pattern in elevation plane. Because this method does not solve any differential equation, it is numerically more efficient and converges faster than the traditional method. The formulation was applied to ADC-like configuration, but can be adapted to other axis displaced dual-reflector configuration.

#### ACKNOWLEDGMENT

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